

**Warsaw University
of Technology**



**Faculty of Power and
Aeronautical Engineering**
WARSAW UNIVERSITY OF TECHNOLOGY

Institute of Aeronautics and Applied Mechanics

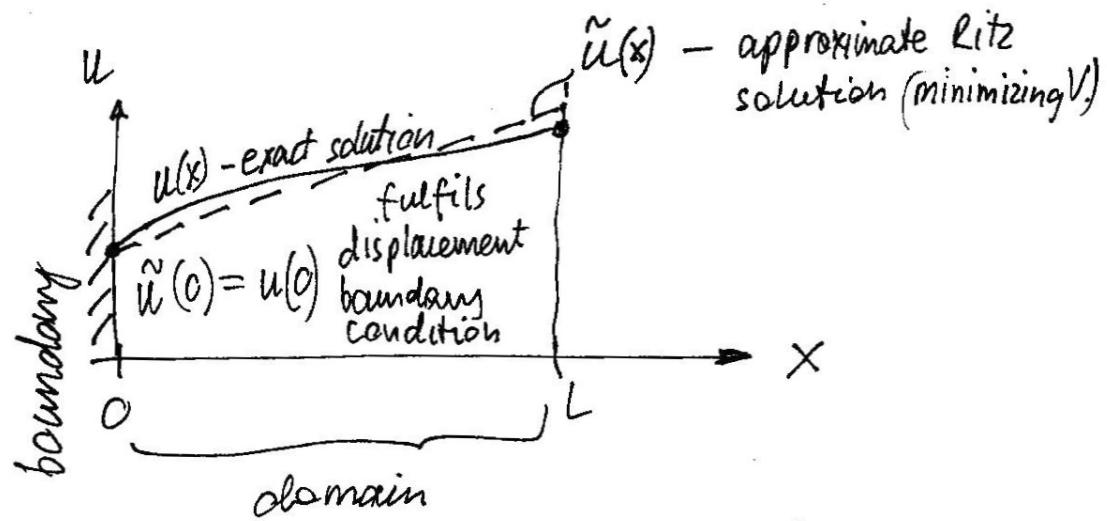
**Finite element method
(FEM)**

Ritz method

04.2021

RITZ METHOD

Ritz method is one of the approximate methods based on the principle of minimum total potential energy and a global approximation.



In this method the approximate function $\tilde{u}(x)$ is assumed for the entire domain as:

$$\tilde{u}(x) = \sum_{i=1}^n c_i \cdot f_i(x)$$

↑ ↑
unknown constants functions

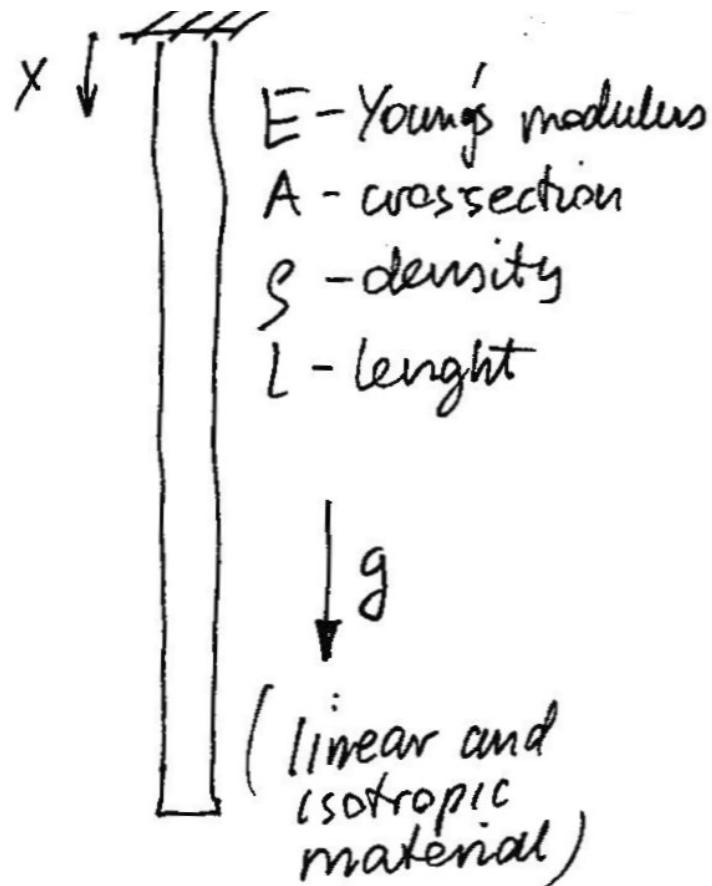
← global approximation

for example :

$$f_i = x^{i-1}; f_1 = 1; f_2 = x; f_3 = x^2; f_4 = x^3; f_5 = x^4; \dots$$

We use a global approximation when the approximate function $\tilde{u}(x)$ is used for the entire domain.

Example. Find displacement, strain and stress in a bar loaded by gravity g . Use the Ritz method ($i=2, 3, 4$). Compare the approximate solution with an exact solution.



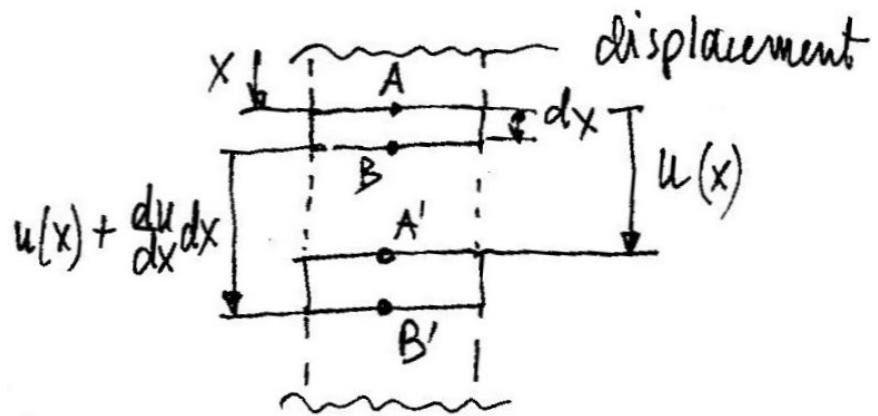
Exact solution

$\uparrow N(0) = \sigma_x(0) \cdot A = m \cdot g$ (reaction)

 equilibrium equation:

$$-\sigma_x \cdot A + \int_x^L \rho g A dx = 0$$

mass force $\sigma_x = \rho g (L-x)$
 $\sigma_x(0) = \rho g L$
 $\sigma_x(L) = 0$
 $dQ = dm \cdot g = dV \cdot \rho g = \rho g A dx$
 traction: $\frac{dQ}{dx} = \rho g A$



$$\epsilon_x = \frac{\Delta x}{E} = \frac{\rho g(l-x)}{E}$$

$$\epsilon_x(0) = \frac{\rho g l}{E}, \quad \epsilon_x(l) = 0$$

$$\begin{aligned} \epsilon_x &= \frac{(A'B') - AB}{AB} = \frac{dx + u(x) + \frac{du}{dx}dx - u(x) - dx}{dx} = \\ &= \frac{du}{dx} \Rightarrow \end{aligned}$$

$$u(x) = \int_0^x \epsilon_x dx + u(0) = \int_0^x \frac{\rho g(l-x)}{E} dx = \frac{\rho g}{E} \left(lx - \frac{x^2}{2} \right)$$

$$u(0) = 0, \quad u\left(\frac{l}{2}\right) = \frac{\rho g}{E} \left(\frac{l^2}{2} - \frac{l^2}{4 \cdot 2} \right) = \frac{3 \rho g l^2}{8 E}, \quad u(l) = \frac{\rho g l^2}{2 E}$$

$$1^0) n=2, \quad \tilde{u}(x) = c_1 \cdot f_1 + c_2 f_2 = c_1 \cdot 1 + c_2 \cdot x$$

$$\text{boundary condition } \tilde{u}(0) = 0 \Rightarrow c_1 + c_2 \cdot 0 = 0 \Rightarrow c_1 = 0$$

$$\tilde{u}(x) = c_2 \cdot x, \quad \tilde{\epsilon}_x = \frac{d\tilde{u}}{dx} = c_2, \quad \tilde{\sigma}_x = E \tilde{\epsilon}_x = E \cdot c_2$$

$$U = \frac{1}{2} \int (\tilde{\sigma}_x \tilde{\epsilon}_x + 0 \cdot \tilde{\epsilon}_y + 0 \cdot \tilde{\epsilon}_z + 0 \cdot 0 + 0 \cdot 0 + 0 \cdot 0) d\Omega =$$

$$= \frac{1}{2} \int_0^L \tilde{\sigma}_x \tilde{\epsilon}_x dA dx = \frac{1}{2} \int_0^L E c_2 \cdot c_2 \cdot A dx = \frac{EA}{2} c_2^2 L$$

$$W = \underbrace{\int_{S2} \sum_{1 \times 3} \sum_{3 \times 1} \{u\} d\Omega}_{+ \underbrace{\int_{\Gamma} \sum_{1 \times 3} \sum_{3 \times 1} \{u\} d\Gamma}_{\text{O (no surface load)}}} = \int_{S2} (X \tilde{u} + 0 \tilde{v} + 0 \tilde{w}) d\Omega =$$

$$= \int_0^L \rho g \tilde{u} \int_A dA dx = \int_0^L \rho g A \tilde{u} dx = \int_0^L \rho g A c_2 x dx = \frac{\rho g A L^2}{2} c_2$$

$$V = \frac{EA}{2} c_2^2 l - \frac{sgAl^2}{2} c_2 ; \quad V \rightarrow \min$$

$$\frac{\partial V}{\partial c_2} = 0 \Rightarrow 2 \frac{EA}{2} c_2 \cdot l - \frac{sgAl^2}{2} = 0$$

$$c_2 = \frac{sgAl^2}{2EAl} = \frac{sgl}{2E}$$

$$\tilde{u}(x) = \frac{sgl}{2E}x ; \quad \tilde{u}(0) = 0 ; \quad \tilde{u}(l) = \frac{sgl^2}{2E}$$

$$\tilde{u}(x) \neq u(x)$$

$$\tilde{u}(l) = u(l)$$

$$\tilde{\epsilon}_x = \frac{sgl}{2E} = \text{const} = \frac{\epsilon_x(0) + \epsilon_x(l)}{2}$$

$$\tilde{\sigma}_x = \frac{sgl}{2} = \text{const} = \frac{\sigma_x(0) + \sigma_x(l)}{2}$$

$$2^\circ) n=3; \tilde{u}(x) = c_1 f_1 + c_2 f_2 + c_3 f_3 = c_1 \cdot 1 + c_2 \cdot x + c_3 \cdot x^2$$

boundary condition: $\tilde{u}(0)=0 \Rightarrow c_1 + c_2 \cdot 0 + c_3 \cdot 0^2 = 0 \Rightarrow c_1 = 0$

$$\tilde{u}(x) = c_2 x + c_3 x^2, \quad \tilde{\epsilon}_x = \frac{d\tilde{u}}{dx} = c_2 + 2c_3 x, \quad \tilde{\sigma}_x = E \cdot \tilde{\epsilon}_x$$

$$U = \frac{1}{2} \int_0^L \tilde{\sigma}_x \tilde{\epsilon}_x A dx = \frac{EA}{2} \int_0^L (c_2 + 2c_3 x)^2 dx =$$

$$= \frac{EA}{2} \int_0^L (c_2^2 + 4c_2 c_3 x + 4c_3^2 x^2) dx =$$

$$= \frac{EA}{2} \left(c_2^2 x + 2c_2 c_3 x^2 + \frac{4}{3} c_3^2 x^3 \right) \Big|_0^L = \frac{EA}{2} \left(L c_2^2 + 2L c_2 c_3 + \frac{4}{3} L^3 c_3^2 \right)$$

$$W = \int_0^L \rho g A \tilde{u} dx = \rho g A \int_0^L (c_2 x + c_3 x^2) dx = \rho g A \left(\frac{c_2 x^2}{2} - \frac{c_3 x^3}{3} \right) \Big|_0^L =$$

$$= \rho g A \left(\frac{L^2}{2} c_2 - \frac{L^3}{3} c_3 \right), \quad V \rightarrow \min$$

$$\begin{cases} \frac{\partial V}{\partial C_2} = 0 \\ \frac{\partial V}{\partial C_3} = 0 \end{cases} \Rightarrow \begin{cases} \frac{\partial U}{\partial C_2} - \frac{\partial W}{\partial C_2} = 0 \\ \frac{\partial U}{\partial C_3} - \frac{\partial W}{\partial C_3} = 0 \end{cases} \Rightarrow \begin{cases} \frac{EA}{2}(2lC_2 + 2l^2C_3) - \frac{89AL^2}{2} = 0 \\ \frac{EA}{2}(2l^2C_2 + \frac{8l^3}{3}C_3) - \frac{89AL^3}{3} = 0 \end{cases}$$

$$\begin{array}{l} (i_1) \left\{ 2lC_2 + 2l^2C_3 = \frac{89L^2}{E} \right. \\ (i_2) \left\{ 2l^2C_2 + \frac{8l^3}{3}C_3 = \frac{289L^2}{3E} \right. \end{array} \Rightarrow \begin{cases} 2C_2 + 2lC_3 = \frac{89L}{E} \\ 2C_2 + \frac{8}{3}lC_3 = \frac{289L}{3E} \end{cases}$$

$$① - ② : -\frac{2}{3}lC_3 = \frac{1}{3}\frac{89L}{E} \Rightarrow C_3 = -\frac{89}{2E}, C_2 = \frac{89L}{2E} - lC_3 = \frac{89L}{E}$$

$$\tilde{u}(x) = \frac{89L}{E}x - \frac{89E}{2E}x^2 = \frac{89}{E}\left(1x - \frac{x^2}{2}\right) = u(x)$$

$$\tilde{\epsilon}_x = \epsilon_x, \tilde{\sigma}_x = \sigma_x$$

$$3^{\circ}) n=4, \tilde{u}(x) = c_1 f_1 + c_2 f_2 + c_3 f_3 + c_4 f_4 = c_1 \cdot 1 + c_2 x + c_3 x^2 + c_4 x^3$$

boundary condition $\tilde{u}(0)=0 \Rightarrow c_1 + 0 + 0 + 0 = 0 \Rightarrow c_1 = 0$

$$\tilde{u}(x) = c_2 x + c_3 x^2 + c_4 x^3, \tilde{\epsilon}_x = \frac{d\tilde{u}}{dx} = c_2 + 2c_3 x + 3c_4 x^2, \tilde{\sigma}_x = E \tilde{\epsilon}_x$$

$$U = \frac{EA}{2} \int_0^L (c_2 + 2c_3 x + 3c_4 x^2)^2 dx$$

$$V = U - W \rightarrow \min$$

$$W = \rho g A \int_0^L (c_2 x + c_3 x^2 + c_4 x^3) dx \quad \frac{\partial V}{\partial c_2} = 0, \frac{\partial V}{\partial c_3} = 0, \frac{\partial V}{\partial c_4} = 0$$

$$\therefore c_2 = \frac{\rho g L}{E}, \quad c_3 = -\frac{\rho g}{2E}, \quad c_4 = 0$$

$$\tilde{u}(x) = \frac{\rho g}{E} \left((x - \frac{x^2}{2}) \right) = u(x)$$

$$\tilde{\epsilon}_x(x) = \epsilon_x(x), \quad \tilde{\sigma}_x(x) = \sigma_x(x)$$

$$4^{\circ}) \quad f_i = \sin \frac{i\pi x}{2L} \quad i=1, 2, \dots n$$

$$n=2 \quad \tilde{w}(x) = c_1 \cdot \sin \frac{\pi x}{2L} + c_2 \cdot \sin \frac{\pi x}{L}$$

$$\tilde{\epsilon}_x = \frac{d\tilde{w}}{dx} = \frac{c_1 \pi}{2L} \cdot \cos \frac{\pi x}{2L} + \frac{c_2 \pi}{L} \cdot \cos \frac{\pi x}{L}, \quad \tilde{\sigma}_x = E \cdot \tilde{\epsilon}_x$$

$$U = \frac{1}{2} \int_{-L}^L \tilde{\sigma}_x \tilde{\epsilon}_x dL = \frac{EA}{2} \int_0^L \tilde{\epsilon}_x^2 dx = \frac{EA}{2} \int_0^L \left(\frac{c_1 \pi}{2L} \cos \frac{\pi x}{2L} + \frac{c_2 \pi}{L} \cos \frac{\pi x}{L} \right)^2 dx = \\ = \frac{EA}{2} \int_0^L \left(\frac{c_1^2 \pi^2}{4L^2} \cdot \cos^2 \frac{\pi x}{2L} + \frac{2c_1 c_2 \pi^2}{2L^2} \cos \frac{\pi x}{2L} \cdot \cos \frac{\pi x}{L} + \frac{c_2^2 \pi^2}{L^2} \cdot \cos^2 \frac{\pi x}{L} \right) dx \Rightarrow$$

$$\frac{EA}{2} \frac{C_1^2 \pi^2}{4L^2} \int_0^L \cos^2 \frac{\pi x}{2L} dx = \left| \begin{array}{l} y = \frac{\pi x}{2L} \\ dy = \frac{\pi}{2L} dx \\ x=0 \Rightarrow y=0 \\ x=L \Rightarrow y=\frac{\pi}{2} \end{array} \right| = \frac{EA C_1^2 \pi}{4L} \int_0^{\frac{\pi}{2}} \cos^2 y dy =$$

$$= \left| \begin{array}{l} \text{by parts} \\ \int f dg = fg - \int g df \\ f = \cos y, dg = \cos y dy \\ g = \int dg = \sin y \\ df = -\sin y dy \end{array} \right| = \frac{EA C_1^2 \pi}{4L} \left(\cos y \sin y \Big|_0^{\frac{\pi}{2}} + \int_0^{\frac{\pi}{2}} \sin^2 y dy \right) =$$

$$= \frac{EA C_1^2 \pi}{4L} \int_0^{\frac{\pi}{2}} (1 - \cos^2 y) dy = \frac{EA C_1^2 \pi^2}{8L} - \frac{EA C_1^2 \pi}{4L} \int_0^{\frac{\pi}{2}} \cos^2 y dy$$

$$\Rightarrow \frac{EA}{2} \frac{C_1^2 \pi^2}{4L^2} \int_0^L \cos^2 \frac{\pi x}{2L} dx = \frac{1}{2} \frac{EA C_1^2 \pi^2}{8L} = \frac{\pi^2 EA}{16L} C_1^2$$

$$\begin{aligned}
 & \frac{EA}{2} \cdot \frac{2c_1 c_2 \pi^2}{2l^2} \int_0^l \cos \frac{\pi x}{2l} \cdot \cos \frac{\pi x}{l} dx = \left| \begin{array}{l} \cos \frac{\pi x}{2l} \cdot \cos \frac{\pi x}{l} = \\ = \frac{1}{2} \left(\cos \left(\frac{\pi x}{2l} - \frac{\pi x}{l} \right) + \cos \left(\frac{\pi x}{2l} + \frac{\pi x}{l} \right) \right) \\ = \frac{1}{2} \left(\cos \frac{-\pi x}{2l} + \cos \frac{3\pi x}{2l} \right) \end{array} \right| = \\
 & = \frac{EAc_1 c_2 \pi^2}{2l^2} \cdot \left(\frac{1}{2} \int_0^l \cos \left(-\frac{\pi x}{2l} \right) dx + \frac{1}{2} \int_0^l \cos \frac{3\pi x}{2l} dx \right) = \\
 & = \frac{EAc_1 c_2 \pi^2}{4l^2} \cdot \left(-\frac{2l}{\pi} \cdot \sin \left(-\frac{\pi x}{2l} \right) \Big|_0^l + \frac{2l}{3\pi} \cdot \sin \frac{3\pi x}{2l} \Big|_0^l \right) = \\
 & = \frac{EAc_1 c_2 \pi^2}{4l^2} \left(-\frac{2l}{\pi} \cdot (-1) + \frac{2l}{3\pi} (-1) \right) = \frac{EAc_1 c_2 \pi^2}{4l} \cdot \frac{2}{3} \cdot \frac{2l}{\pi} = \frac{\pi EA}{3l} c_1 \cdot c_2
 \end{aligned}$$

$$\frac{EA}{2} \frac{c_2^2 \pi^2}{l^2} \int_0^l \cos^2 \frac{\pi x}{l} dx = \frac{\pi^2 EA}{4l} c_2^2$$

$$U = \frac{\pi^2 EA}{16l} c_1^2 + \frac{\pi EA}{3l} c_1 \cdot c_2 + \frac{\pi^2 EA}{4l} c_2^2$$

$$\begin{aligned}
 W &= \int_R \rho g \tilde{u}(x) dx = \rho g A \int_0^L (c_1 \cdot \sin \frac{\pi x}{2L} + c_2 \cdot \sin \frac{\pi x}{L}) dx = \\
 &= \rho g A c_1 \int_0^L \sin \frac{\pi x}{2L} dx + \rho g A c_2 \int_0^L \sin \frac{\pi x}{L} dx = \\
 &= 2 \rho g A c_1 L \left(-\cos \frac{\pi x}{2L} \right) \Big|_0^L + \rho g A c_2 \frac{L}{\pi} \left(-\cos \frac{\pi x}{L} \right) \Big|_0^L = \\
 &= \underline{\underline{\frac{2 \rho g A L}{\pi} \cdot c_1}} + \underline{\underline{\frac{2 \rho g A L}{\pi} \cdot c_2}}
 \end{aligned}$$

$$V = \frac{\pi^2 EA}{16L} c_1^2 + \frac{EA\pi}{3L} c_1 c_2 + \frac{\pi^2 EA}{4L} c_2^2 - \frac{289AL}{\pi} (c_1 + c_2)$$

$$\frac{\partial V}{\partial c_1} = 0 \Rightarrow \left\{ \begin{array}{l} \frac{\pi^2 EA}{8L} c_1 + \frac{EA\pi}{3L} c_2 - \frac{289AL}{\pi} = 0 \end{array} \right.$$

$$\frac{\partial V}{\partial c_2} = 0 \Rightarrow \left\{ \begin{array}{l} \frac{EA\pi}{3L} \cdot c_1 + \frac{\pi^2 EA}{2L} c_2 - \frac{289AL}{\pi} = 0 \\ \cdot \frac{L}{EA\pi} \end{array} \right.$$

$$\begin{bmatrix} \frac{\pi}{8} & \frac{1}{3} \\ \frac{1}{3} & \frac{\pi}{2} \end{bmatrix} \cdot \begin{Bmatrix} c_1 \\ c_2 \end{Bmatrix} = \begin{Bmatrix} \frac{289L^2}{E\pi^2} \\ \frac{289L^2}{E\pi^2} \end{Bmatrix}$$

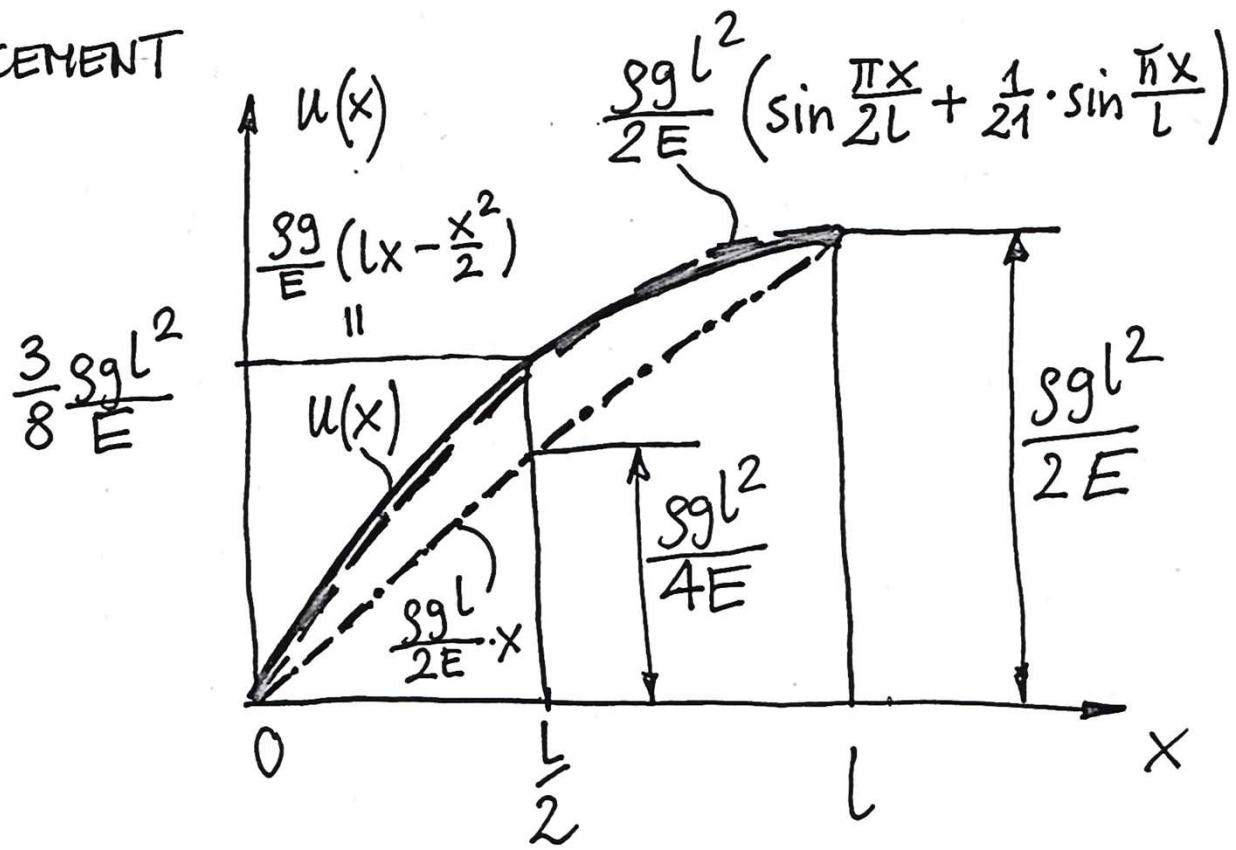
$$\begin{Bmatrix} c_1 \\ c_2 \end{Bmatrix} = \begin{bmatrix} \frac{\pi}{8} & \frac{1}{3} \\ \frac{1}{3} & \frac{\pi}{2} \end{bmatrix}^{-1} \cdot \frac{289l^2}{E\pi^2} \begin{Bmatrix} 1 \\ 1 \end{Bmatrix}$$

$$c_1 \approx \frac{89l^2}{2E}, \quad c_2 \approx \frac{89l^2}{42E}$$

$$\tilde{u}(x) = \frac{89l^2}{2E} \left(\sin \frac{\pi x}{2l} + \frac{1}{21} \sin \frac{7\pi x}{l} \right)$$

$$\tilde{e}_x(x) = \frac{d\tilde{u}}{dx} = \frac{\pi 89l}{4E} \left(\cos \frac{\pi x}{2l} + \frac{2}{21} \cos \frac{7\pi x}{l} \right)$$

DISPLACEMENT

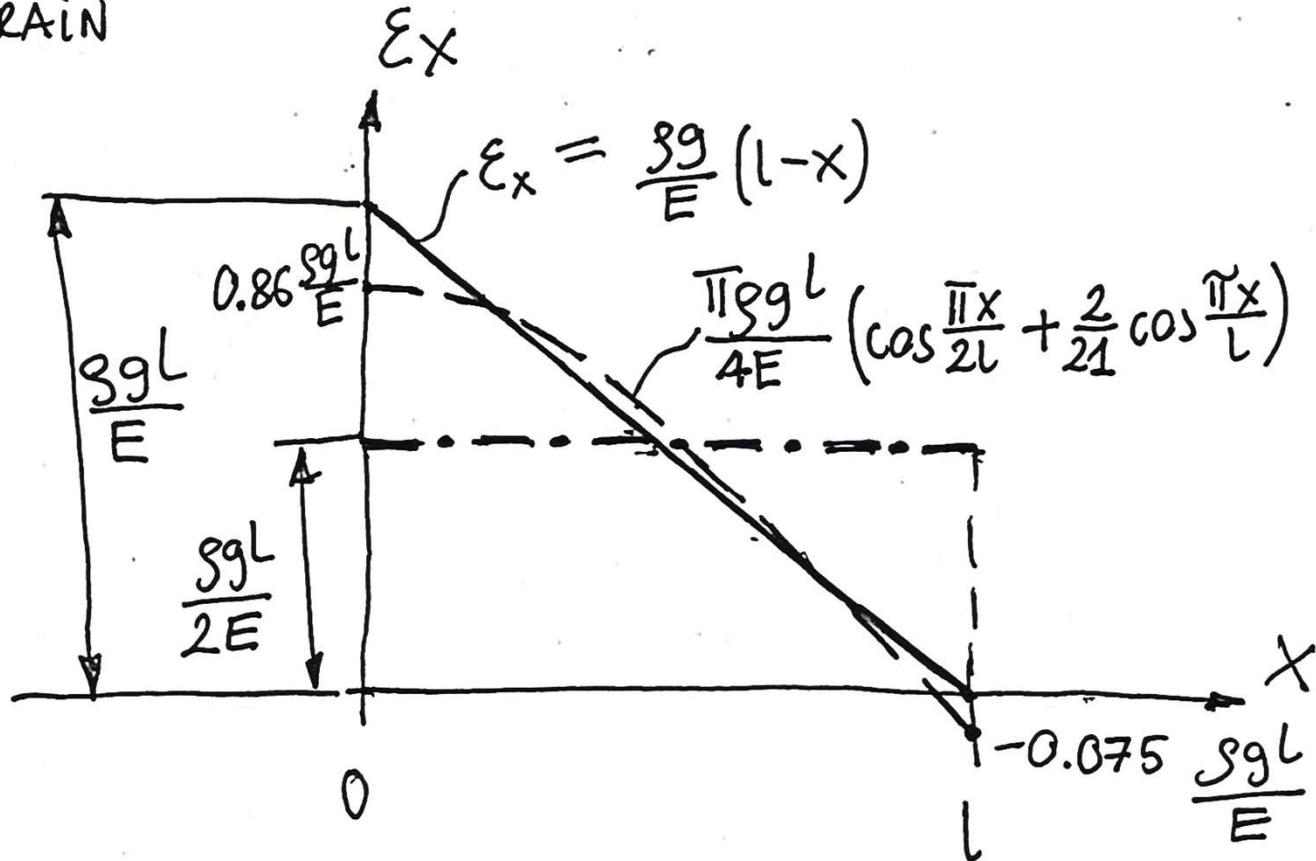


—·—·— Ritz $n=2$ (linear function)

— — — exact solution , Ritz $n=3,4,\dots$

— — — Ritz $n=2$, sine functions

STRAIN



—·—·— Ritz $n=2$ (linear function)

— — — exact solution, Ritz $n=3, 4, \dots$

— — — Ritz $n=2$, sine functions